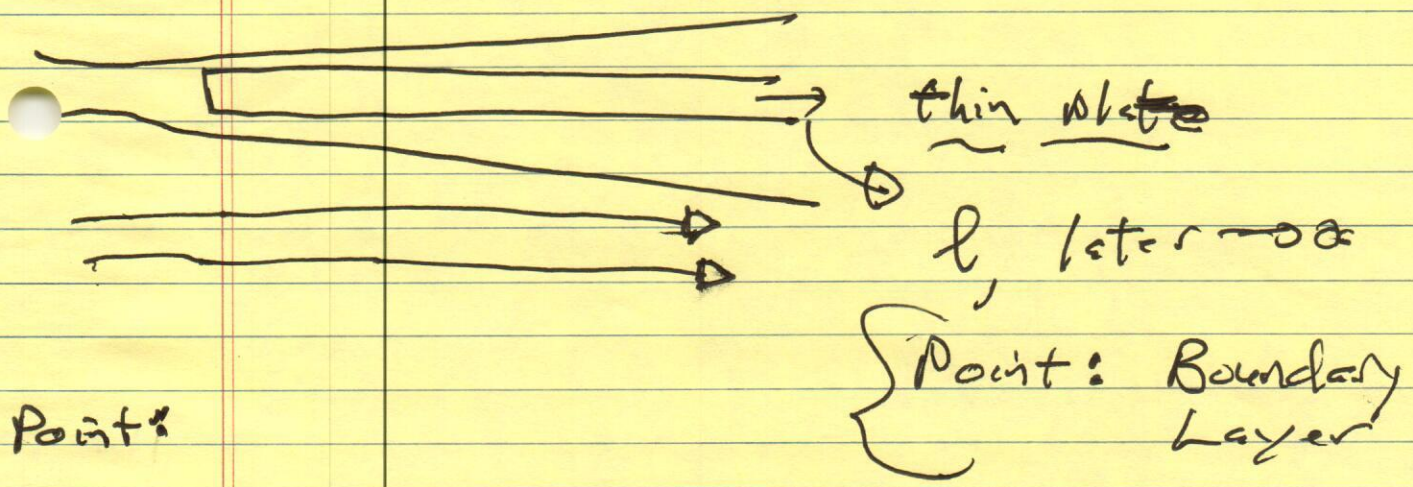


Lecture VIII - Laminar: Boundary Layers, Wakes

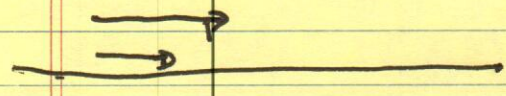
N.B.: For stability, see also:
P. Drazin, W. H. Reid: "Hydrodynamic Stability"

Here: - Laminar BL $\rightarrow Re \gg 1$, but not turbulent
- Laminar Wake - see Lect. 7.
(L.L. / Falkovich)

(Blasius) Boundary Layer Problem
Potential flow



Point:



$$v_n = 0 \mid p$$

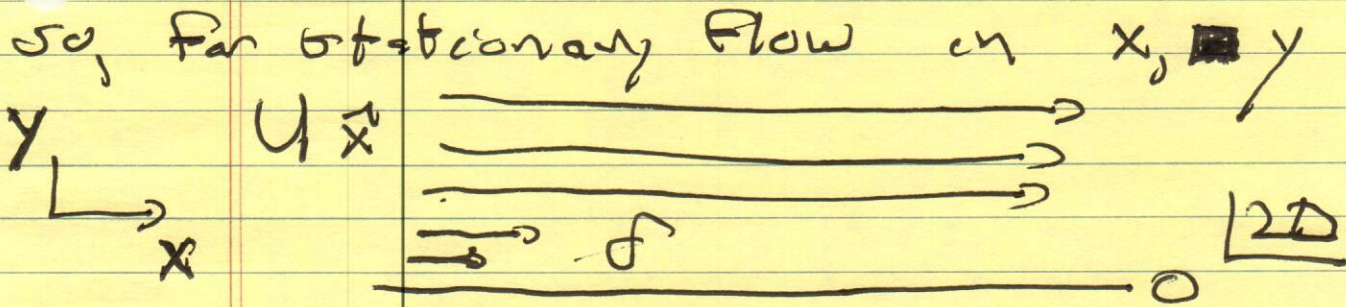
$$v_t = 0 \mid p$$

- boundary layer forms
in flow as consequence
of higher Re .

$Re \gg 1$

layer is 'thin', as Re large.

- contrast Stokes' dynamics \rightarrow
viscosity dominant everywhere



$$\partial_x v_x + \partial_y v_y = 0$$

$$v_x \partial_x v_x + v_y \partial_y v_x = -\frac{1}{\rho} \partial_x p + \nu (\partial_x^2 v_x + \partial_y^2 v_x)$$

$$v_x \partial_x v_y + v_y \partial_y v_y = -\frac{1}{\rho} \partial_y p + \nu (\partial_x^2 v_y + \partial_y^2 v_y)$$

Now, further simplify:

→ v_y varies rapidly in y
i.e. on scales of the BL thickness

$$\Rightarrow \underline{\underline{\partial_x^2 v_x}} \ll |\partial_y^2 v_x|$$

i.e. along plate diffn negligible

→ similarly, $|v_x| > |v_y|$
and $|\partial_x p| \gg |\partial_y p|$

con

- streamlines flat

- $|v_y| \ll |v_x|$

so

$$\rho(x) + \frac{1}{2} \rho U(x)^2 = \text{const.}$$

Pressure balance also Bernoulli, \vec{x} direction

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$$v_x \partial_x v_x + v_y \partial_y v_x = -\frac{1}{\rho} \partial_x p + \nu \partial_y^2 v_x$$

$$\partial_x v_x + \partial_y v_y = 0$$

$$\text{with: } \partial_x p = -\rho U \partial_x U$$

Prandtl equations.

with:

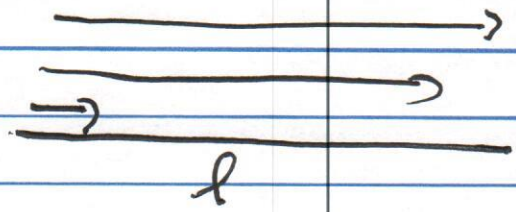
$$- \text{b.c. } v_x = v_y = 0 \text{ for } y = 0$$

$$- v_x = U(x) \text{ for } y \rightarrow \infty$$

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- can be generalized to other 2D flow.

Now



Proceed by
BOE

- U_0 is flow as $y \rightarrow \infty$

- at BOE level:

$$\underbrace{u_x \partial_x u_x + v_y \partial_y u_x}_{\text{(same)}} = \underbrace{-\frac{1}{\rho} \partial_x p}_{\text{ignore}} + \nu \underbrace{\partial_y^2 u_x}_{\frac{\nu}{\delta^2} u_x}$$

$$\partial_x u_x + \partial_y u_y = 0$$

$$\frac{u_x}{l} \sim \frac{u_y}{\delta}$$

suggests variable change.

So

$$\frac{u_p u_x}{l} \sim \frac{\nu}{\delta^2} u_x$$

$$\delta \sim \left(\frac{l \nu}{U_0} \right)^{1/2} \rightarrow \text{Blasius BL thickness}$$

$$\rightarrow \frac{l}{Re} \text{ plate scale.}$$

In case you are wondering:

$$\gamma: \frac{\rho}{\rho_0} \sim \frac{v_x v_y}{d^2}$$

(if assume, ρ not negligible)

$$\frac{v_x}{l} \sim \frac{v_y}{d}$$

$$\rho \sim \rho_0 v_x / l$$

$$\frac{\rho}{\rho_0} \sim \frac{v_x v_y}{d^2} \ll \frac{v_x}{d}$$

$\frac{\rho}{\rho_0}$ negligible!

$$Re \sim U l / \nu$$

suggests:

$$x = l x', \quad y = l y' / \sqrt{Re}$$

$$U_x = U_0 u_x', \quad U_y = U_0 l u_y' / \sqrt{Re}$$

$$U = U_0 U'$$

so, rescaled Prandtl eqns: (dim-less)

$$u_x' \frac{\partial u_x'}{\partial x'} + u_y' \frac{\partial u_x'}{\partial y'} - \frac{\partial^2 u_x'}{\partial y'^2} = U_0 l \frac{U'}{l} \frac{\partial U'}{\partial x'}$$

$$\frac{\partial u_x'}{\partial x'} + \frac{\partial u_y'}{\partial y'} = 0$$

Note: = Re scaled out
- change in Re \Rightarrow rescaling
- x unchanged
y $1/\sqrt{Re}$
Reynolds # similarity

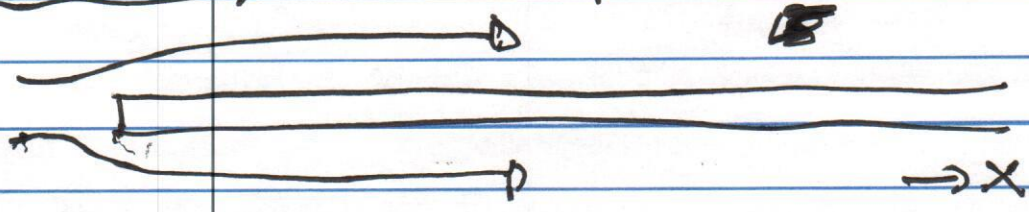
thus, immediately:

$$\delta \sim l / \sqrt{Re}$$

$$U_y \sim U / \sqrt{Re}$$

now, can specify to Blasius Problem \Rightarrow semi-infinite plate.

Semi-infinite plate is rigorously solvable by ~~similarity~~ methods.



- here: expect δ thickens with x as characteristic scale gone.

c.e. $\delta^2 \sim \nu x$
 $\sim \nu x / U$
 $\Rightarrow \delta \sim (\nu x / U)^{1/2}$

\downarrow
 Blasius B.L. thickness

- $U = U(x) \rightarrow \text{const } U_0$

so

$$v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_y}{\partial y} = \nu \frac{\partial^2 v_x}{\partial y^2} + \frac{\rho}{\mu} \frac{\partial U^2}{\partial x}$$

$$\frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} = 0$$

Now, no characteristic length

so

where before $x' = x/l$
 $y' = y (U/\nu)^{1/2}$

Ref. on singularity: G.I. Barenblatt
 "Scaling"

- Now, need get δ out of problem

so

$$- v_x/u = F(y'/\sqrt{x'}) \quad \frac{y'}{\sqrt{x'}} = y' \sqrt{u/\nu x}$$

$v(x,y) \rightarrow F(y'/\sqrt{x'}) \rightarrow$ self-sim. \rightarrow simplest possible not involving ρ

$$- v'_y \sqrt{x'} = G(y'/\sqrt{x'})$$

so $y'/\sqrt{x'} \sim \Sigma \rightarrow$ self-similarity variable

$$\Rightarrow \delta \sim (\nu x/u)^{1/2} \quad \text{BL thickness}$$

To exploit:

$$\underline{v} = \begin{pmatrix} \frac{\partial \psi}{\partial y} & -\frac{\partial \psi}{\partial x} \end{pmatrix} \quad \text{cls: } \underline{\partial \psi x \bar{z}}$$

\downarrow to make $x \rightarrow y$

$$\psi \equiv (\nu x u)^{1/2} f(\Sigma) \quad \text{self-similarity variable}$$

$$\Sigma = y'/\sqrt{x'} \rightarrow y (u/\nu x)^{1/2}$$

again: $\delta \sim (\nu x / U)^{1/2}$

Now

$$Ux \frac{\partial U_x}{\partial x} + \nu y \frac{\partial U_y}{\partial y} = \nu \frac{\partial^2 U_x}{\partial y^2}$$

$$\psi = (\nu x U)^{1/2} F(\xi)$$

$$\xi = y (\nu / \nu x)^{1/2}$$

$U_x = \partial \psi / \partial y$
 $U_y = -\partial \psi / \partial x$

Plug in:

$$F F'' + 2 F''' = 0$$

Num. similarity variable
 PDE \rightarrow ODE reduces

$f(0) = f'(0) = 0, f'(\infty) = 1$ B.C

and can solve, { numeric, some asymptotic } see L2

$$U_x = U F'(\xi)$$

$$U_y = \frac{1}{2} \sqrt{\nu U / x} (\xi F' - F)$$

→ More useful to estimate!

$$\tau_{xy} = \eta \frac{\partial u_x}{\partial y} \sim \eta \frac{U}{\delta}$$

frictional force/area (stress) on plate

$$\sim \eta \left(\frac{U^2}{\nu x} \right)^{1/2} \underbrace{f''(0)}$$

$$\rightarrow 0.332 (\eta \rho U^3 / x)^{1/2}$$

IF plate has length l ,

$$\frac{F}{L} = \int_0^l dx \eta \frac{\partial u_x}{\partial y} \sim \int_0^l dx \frac{\eta U}{(\nu x / U)^{1/2}}$$

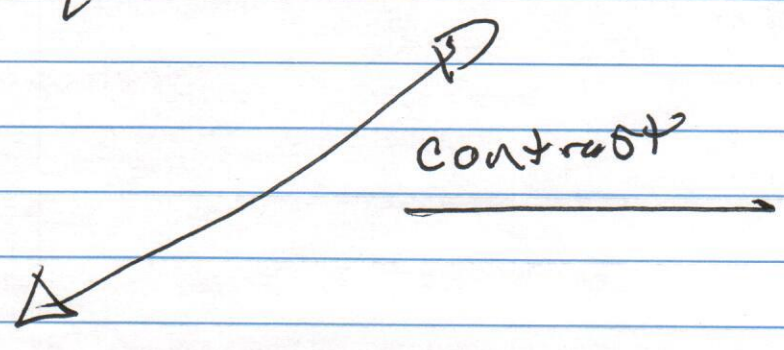
force/length

$$\sim (\eta \rho U^3)^{1/2}$$

$$\sim \nu^{1/2} l^{1/2} \eta^{1/2} \Rightarrow F \sim \nu^{3/2} l^{3/2} U^{1/2}$$

Contrast Stokes:
 $6\pi \eta U^0$

$$F \sim U l \eta$$



and, define drag coefficient

C_d via: defn.

$$C_d = \left[\frac{F}{L} / \frac{1}{2} \rho U^2 \right] 2l$$

2D defn.

↳ for 2D flow.

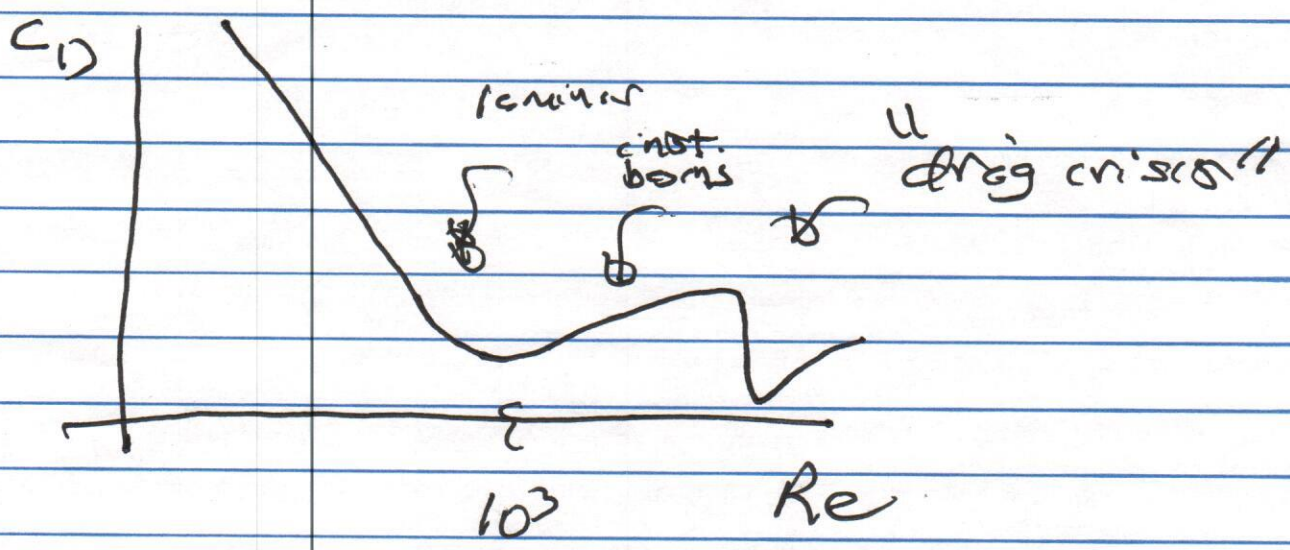
in general

~ # / \sqrt{Re}

$$C_d = \frac{F}{\frac{1}{2} \rho U^2 S}$$

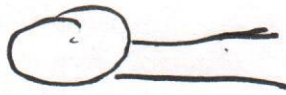
↓
surface area

Comments on D_{reg} :



→ $Re > 1 \Rightarrow$ laminar BL

→ $Re > 10^3 \Rightarrow$ instability, separation begin, reach body.
 C_D rises



→ turbulence onset ⇒ C_D drops
(drag crisis)

~ B.L. energized

{ why golf balls
have dimples

{ B.L. flow with turb.
can beat viscous
losses.

→ C_D indep. Re ⇒ turbulence physics

Wakes

For laminar wake - see Lecture 7.

Point of wake: Viscosity changes global structure of flow.

C.f.: "Finnegans Wake"

by James Joyce.

- ⇒ classical Joyce novel
- incomprehensible
- origin of "quark" (word):
(3 quarks for Master Mark)

⇒ not related to fluid dynamics except possibly beer consumption.